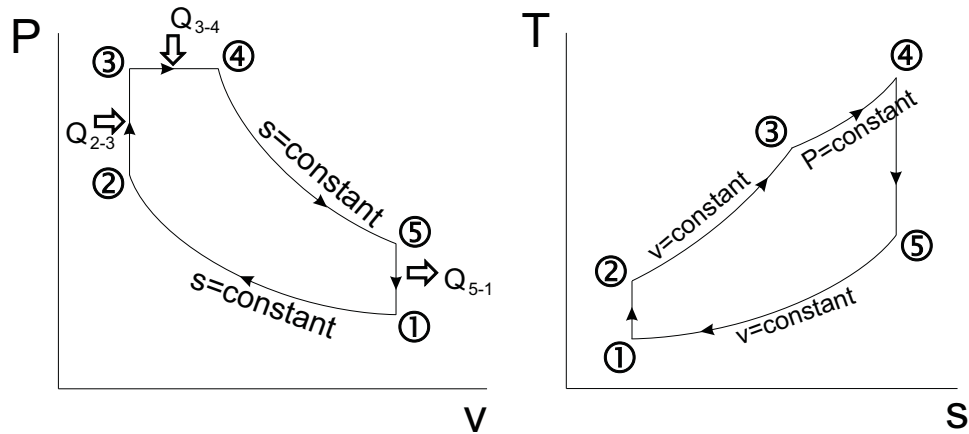


Part a)



State	$P$ (kPa)	$T$ (K)
1	100	293.15
2	2511.9	735.98
3	10434.36	3057.24
4	10434.36	3886.43
5	1326.44	1703.10

**Process 1 - 2**

The process is isentropic so

$$Pv^k = \text{const}$$

where  $k = 1.4$ 

We know from the isentropic equations

$$\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{(k-1)/k}$$

and

$$\frac{P_2}{P_1} = \left(\frac{V_1}{V_2}\right)^k = (10)^{1.4} = 25.119$$

Then

$$P_2 = (100 \text{ kPa})(25.119) = 2511.9 \text{ kPa} \Leftarrow$$

$$T_2 = T_1 \left(\frac{V_1}{V_2}\right)^{k-1} = (20 + 273)(10)^{0.4} = 735.98 \text{ K} \Leftarrow$$

**Process 2 - 3**

the volume is constant during this process. Performing a 1st law balance gives

$$q_{2-3} = c_v(T_3 - T_2)$$

At 300 K we know that  $c_v = 0.718 \text{ kJ/kg} \cdot \text{K}$  for air. The problem states that  $2/3$  of the total heat input occurs during this phase of the cycle.

$$\left(\frac{2}{3}\right) 2500 \text{ kJ/kg} = (0.718 \text{ kJ/kg} \cdot \text{K})(T_3 - 735.98)$$

$$T_3 = 3057.24 \text{ K} \Leftarrow$$

For an ideal gas (with constant mass) we know

$$\frac{P_3 V_3}{T_3} = \frac{P_2 V_2}{T_2}$$

But  $V_2 = V_3$ , so

$$\frac{P_3}{T_3} = \frac{P_2}{T_2}$$

$$P_3 = T_3 \frac{P_2}{T_2} = (3057.24 \text{ K}) \left( \frac{2511.9 \text{ kPa}}{735.98 \text{ K}} \right) = 10434.36 \text{ kPa} \Leftarrow$$

**Process 3 - 4**

constant pressure process. We know for this case that

$$q_{3-4} = c_p(T_4 - T_3)$$

Therefore

$$\left(\frac{1}{3}\right) 2500 \text{ kJ/kg} = (1.005 \text{ kJ/kg} \cdot \text{K})(T_4 - 3057.24)$$

$$T_4 = 3886.43 \text{ K} \Leftarrow$$

We know

$$P_4 = P_3 = 10434.36 \text{ kPa} \Leftarrow$$

### Process 4 - 5

This is an isentropic process where

$$\frac{P_5}{P_4} = \left( \frac{V_4}{V_5} \right)^k$$

and it can be shown that

$$\frac{P_5}{P_4} = \left( \frac{V_2}{V_1} \frac{T_4}{T_3} \right)^k$$

or

$$P_5 = (10434.36 \text{ kPa}) \left( \frac{1}{10} \frac{3886.43 \text{ K}}{3057.24 \text{ K}} \right) = 1326.44 \text{ kPa}$$

and

$$T_5 = T_4 \left( \frac{P_5}{P_4} \right)^{(k-1)/k} = 3886.43 \left( \frac{581.27}{10434.36} \right)^{0.4/1.4} = 1703.10 \text{ K}$$

### Part b)

The thermal efficiency is given as

$$\begin{aligned} \eta &= \frac{\text{net work done}}{\text{total heat in}} \\ &= \frac{w_{1-2} + w_{3-4} + w_{4-5}}{q_{in}} \end{aligned}$$

$$w_{1-2} = -c_v(T_2 - T_1) = -(0.718 \text{ kJ/kg} \cdot \text{K})(735.98 \text{ K} - 293 \text{ K}) = -318.06 \text{ kJ/kg}$$

$$\begin{aligned} w_{3-4} &= q_{3-4} - c_v(T_4 - T_3) \\ &= \left( \frac{1}{3} \right) (2500 \text{ kJ/kg}) - (0.718 \text{ kJ/kg} \cdot \text{K})(3886.43 \text{ K} - 3057.24 \text{ K}) \\ &= 237.97 \text{ kJ/kg} \end{aligned}$$

$$w_{4-5} = -c_v(T_5 - T_4) = -(0.718 \text{ kJ/kg} \cdot \text{K})(1703.1 \text{ K} - 3886.43 \text{ K}) = 1567.63 \text{ kJ/kg}$$

Therefore the efficiency is

$$\eta = \frac{-318.06 + 237.97 + 1567.63}{2500} = 59.5\%$$

Part c)

We know that  $V_2 = V_3$  and  $V_1 = V_5$  and for the isentropic process between 4 – 5

$$\frac{V_4}{V_5} = \frac{V_4}{V_1} = \left(\frac{P_5}{P_4}\right)^{1/k}$$

and

$$r = \frac{V_1}{V_2} = \frac{V_1}{V_3}$$

Therefore the cutoff ratio is

$$r_v = \frac{V_4}{V_3} = \frac{V_1(P_5/P_4)^{1/k}}{V_1/r} = (P_5/P_4)^{1/k} \cdot r = 10 \cdot \left(\frac{581.27}{10434.36}\right)^{1/1.4} = 1.27$$

$$r_p = \frac{P_3}{P_2} = \frac{10434.36}{2511.9} = 4.154$$

$$\begin{aligned}\eta &= 1 - \frac{r_p r_v^k - 1}{[(r_p - 1) + k r_p (r_v - 1)] r^{k-1}} \\ &= 1 - \frac{(4.154)(1.27)^{1.4} - 1}{[(4.154 - 1) + (1.4)(4.154)(1.27 - 1)](10)^{0.4}} \\ &= 0.595 \leftarrow\end{aligned}$$

Using the isentropic relationships,  $T_{2s}$  can be calculated as

$$T_{2s} = T_1 \times (r_p)^{k-1/k} = (25 + 273) \times (18)^{0.4/1.4} = 680.56 \text{ K}$$

Since the compressor efficiency is 100%,  $T_2$  is the same as  $T_{2s}$

$$T_2 = T_{2s} = 680.56 \text{ K}$$

Using the isentropic relationship for the turbine

$$T_{4s} = T_3 \times \left(\frac{1}{r_p}\right)^{k-1/k} = (1000 + 273) \times \left(\frac{1}{18}\right)^{0.4/1.4} = 557.42 \text{ K}$$

Since the mass flow rate and the specific heat are constant, the isentropic efficiency of the turbine can be written as

$$\begin{aligned} \eta_T &= \frac{T_3 - T_4}{T_3 - T_{4s}} \\ 0.85 &= \frac{1273 - T_4}{1273 - 557.42} \end{aligned}$$

Therefore

$$T_4 = 1273 - 0.85(1273 - 557.42) = 664.76 \text{ K}$$

The net work output per unit mass is

$$\begin{aligned} \frac{W_{net}}{m} &= c_p(T_3 - T_4) - c_p(T_2 - T_1) \\ &= 1.004 \text{ kJ/kg} \cdot \text{K} [(1273 - 664.76) - (680.56 - 298)] \text{ K} \\ &= 226.58 \text{ kJ/kg} \end{aligned}$$

Part a)

The net power output is

$$Power = \dot{m} \times \frac{W_{net}}{m} = 10 \text{ kg/s} \times 226.58 \text{ kJ/kg} = 2265.8 \text{ kW} \Leftarrow$$

Part b)

The thermal efficiency of the engine is

$$\begin{aligned} \eta &= \frac{W_{net}}{Q_{in}} = \frac{T_3 - T_4 - T_2 + T_1}{T_3 - T_2} \\ &= \frac{1273 - 664.76 - 680.56 + 298}{1273 - 680.56} \\ &= 0.381 = 38.1\% \Leftarrow \end{aligned}$$

Note:  $\eta = 1 - (r_p)^{(1-k)/k}$  can only be used for the ideal Brayton cycle where  $T_4 = T_{4s}$